Meandering Riblets Targeting Spanwise Spatial Oscillation of Turbulent Boundary Layers

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Abstract

Turbulent boundary layers are investigated over spanwise oscillating and straight riblets for a range of Reynolds numbers $(1400 < Re_{\tau} < 2800)$. This work is motivated by previous studies of riblets [2] and temporal spanwise oscillation [10] that have both separately demonstrated viscous drag reduction in turbulent boundary layers. Mean velocity profiles acquired over these surfaces are regression fitted to the canonical turbulent boundary layer profile using the roughness modified Clauser and velocity defect plots to determine the friction velocity U_{τ} and virtual origin z_0 . This method for meandering riblets is inconclusive with the variation of U_{τ} and z_0 within the margin of experimental error. For both meandering and straight riblets, robust modifications are observed in the turbulence intensity of the streamwise velocity signal (u') and pre-multiplied energy spectrum $(k_x \phi_{uu})$. A reduction in the near-wall peak of u' is observed for both riblet cases compared to the smooth wall. This is more pronounced for the meandering case. The measured energy spectra in the near-wall region suggest that for the riblet cases the energy contribution from scales consistent with the near-wall cycle are reduced. This is again more pronounced for the meandering case. Finally, it is noted that compared to the smooth wall the meandering riblets increase the magnitude of large-scale turbulent energy in the outer part of the boundary layer $(z/\delta \approx 0.07)$, suggesting that these surfaces modify the largest scale coherent motions residing in the log and wake regions of the flow.

Introduction

Straight riblets have been researched extensively based on their ability to reduce the skin friction of turbulent boundary layers. Reference [8] provides a review of research on straight riblets. Riblets of different cross sectional geometries have been thoroughly investigated by [1]. Riblets of spacing $s^+ = 15$ to 25 and with spacing to height ratio $s/h \approx 0.5$ yield optimal drag reductions (here s is the spanwise peak-to-peak spacing of the riblets, h is the peak-to-trough riblet height and the superscript + represents scaling with viscous units, i.e. $s^+ = sU_{\tau}/\nu$ where U_{τ} is the friction velocity and v is the kinematic viscosity). Active perturbations with spanwise oscillation of wall flows have also been studied for flow control and can yield a drag reduction as high as 40% in turbulent boundary layer flows [7, 10]. However, an energy input is required for the wall oscillation, which when accounted for reduces the net energy savings. The promising aspects of spanwise oscillation for drag reduction coupled with the impracticality of wall oscillations in real world applications has instigated the study of meandering riblets to passively induce spanwise oscillations of turbulent boundary layers. A previous LES study on meandering riblets at $Re_{\tau} = 180$ has been conducted by [9] who obtained a drag reduction of 7.4% (a reported 2% improvement over conventional straight riblets). The work reported in this paper focuses on an experimental study of meandering riblets at moderately high Reynolds Number.

Method

Boundary layer profiles over the meandering and straight riblet

tiles are acquired at Reynolds number Re_{τ} = 1400, 2000 and 2800 (where Re_{τ} is the friction Reynolds number defined as $Re_{\tau} = \delta U_{\tau} / v$ where δ is the boundary layer thickness based on 99.5% of freestream velocity). A single-normal 5 µm hot-wire is mounted on a traverse located 4 m downstream of the trip in the tunnel working section $(0.94 \times 0.375 \text{ m cross section } \times 6.7 \text{ m cross } \times 6.7 \text{ m cross section } \times 6.7 \text{ m cross } \times 6.$ m streamwise length). The hotwire is operated in constant temperature mode using an in-house Melbourne University constant temperature anemometer (MUCTA). The riblets tiles are sized at 500 mm \times 300 mm, with measurements conducted over a test surface consisting of 8 tiles covering a central streamwise strip of length 4 m from the trip inlet in the tunnel. Figure 1 shows the meandering riblet geometry used in the current investigation. Throughout this paper x, y and z refer to the streamwise, spanwise and wall-normal directions respectively. The associated velocity components are u, v and w. Capitalised variables and over bars denote time-averaged values, and lowercase denotes fluctuating quantities.



Figure 1: Schematic of meandering riblets.

Meandering Riblets

Aside from the riblet cross-sectional geometries (h and s) two additional parameters of the meandering riblets are to be determined — the streamwise wavelength Λ_x and amplitude A of the meandering arrangement, as illustrated in Figure 1. Most present literature on spanwise wall oscillations focus on temporal forcing of the oscillating wall. However, [10] performed a DNS study investigating spatial spanwise forcing functions which were reported to yield similar optimal drag reductions. They reported a maximum drag reduction of 52% with a streamwise forcing wavelength $\Lambda_x^+ = 1250$ and spanwise velocity amplitude $V^+ = 20$. Our initial experiment focuses on $Re_{\tau} = 2000$, where the meandering riblets were designed with $\Lambda_x^+ = 1250$ and a streamwise wave amplitude $A^+ = 55$ at this speed. Note that A here is a length scale, whereas V as reported in [10] is a velocity scale. From [10], their results suggests that higher V yields higher drag reduction, with no obvious limiting bound. In contrast, the results of reference [4] suggest that a 15° riblet yaw angle is a limiting bound for drag reduction. The amplitude of the meandering riblets in this study is hence limited such that the maximum yaw angle from the streamwise direction (at the position $\Lambda_x/2$ of a sine wave) is $\theta = 15^\circ$. With a maximum yaw angle of $\theta = 15^\circ$ and a streamwise wavelength of $\Lambda_x = 1250$, we use equation (1) to obtain the approximate maximum spanwise displacement (A^+) of the meandering riblets.

$$A^{+} = \frac{\Lambda_{x}^{+} tan(\theta)}{2\pi} \tag{1}$$

Assuming a convection velocity at the crest of the riblets of approximately $U_c^+ \approx 10$, and assuming the riblets redirect the flow at the meandering angle, we estimate $V^+ \approx 3$ (which [10] shows for $\Lambda_x^+ = 1250$ could give up to 15% drag reductions). The riblets are of 60° triangular cross-section with height and spacing set at $h^+ = 18$ and $s^+ = 25$, which was inherited from previous studies of converging-diverging riblet geometries conducted using the same facilities (see [6] for description).

Straight Riblets

Straight (non-meandering) riblets were also studied to serve as a baseline case to isolate the effect of the meandering arrangement on the boundary layer profiles. The riblet cross-section differs slightly from the meandering case. They are of scallop / semi-circular shaped with height and spacing set at $h^+ = 9$ and $s^+ = 18$, determined from the optimum straight riblet geometries for drag reduction as reported by [1]. Similar manufacturing processes, materials and experimental set-up were adopted as with the meandering riblets. Since the effect of straight riblets is largely confined to the near-wall region [2], it is believed that, it is unlikely that this slight difference in cross-section can account for any differences between the straight and meandering cases further from the surface. Any wider modifications to the large-scale structure inhabiting the outer region of the flow would most likely be attributable to the meandering profile.

Table 1 tabulates the riblet cross-sectional geometries and meandering parameters corresponding to the different Reynolds number experiments. It should be noted that due to difficulties in determining the friction velocity U_{τ} over the ribbed surfaces, all dimensions are here non-dimensionalised using the U_{τ} of smooth wall at that particular Reynolds number.

	$Re_{\tau} = 1400$	$Re_{\tau} = 2000$	$Re_{\tau} = 2800$
$U_{\infty} (\mathrm{ms}^{-1})$	10	15	20
<i>x</i> (m)	4	4	4
Meandering	60° tip triangular cross-section		
h_m^+	12.5	18.0	24.0
s_m^+	17.0	25.0	32.5
Λ_x^+	880	1250	1680
A^+	37.5	55.0	72.0
Straight	scallop/semi circular cross section		
h_s^+	6.0	9.0	12.0
s_s^+	12.0	18.0	24.0

Table 1: Geometries of riblets in wall units. Subscripts *m* and *s* corresponds to meandering and straight riblets accordingly.

Determining the wall-normal position

Experiments are conducted in a zero pressure gradient windtunnel with a working section of $0.94 \text{ m} \times 0.375 \text{ m}$ cross-section and length 6.7 m. The hot-wire probe is placed 4 m downstream from the tripped inlet and is mounted to a cylindrical sting that is attached to a stepper motor driven vertical traverse. A vertically traversing microscope is used to position the probe as close as 0.25 mm from the smooth wall or the riblet tips for the start of the traverse. A camera located outside of the tunnel, positioned 0.5 m away from the probe in the spanwise direction is used to capture any movement of the probe after the tunnel is switched on. Such movements would include any deflection of the cylindrical sting due to aerodynamic loading and also any deflection of the wall of the tunnel due to the tunnel being at positive pressure when in operation. High-resolution images of the hot-wire probe are taken before and after the tunnel is switched on, and any movements are approximated using cross-correlation of the images. We estimate (based on the resolution of the images and repeatability) that an accuracy of $50\mu m$ can be obtained with this technique. The accuracy of the system can be verified by comparing the measured smooth wall mean velocity profile to that obtained from direct numerical simulation (DNS).

Results

For this section, the boundary layer over the meandering and straight riblets are compared with the smooth wall case in several aspects including mean velocity profile, turbulence intensity and premultiplied energy spectrum.

Mean Velocity and Turbulence Intensity

To investigate the presence of drag reduction, we attempt to fit both the smooth and riblet mean velocity profiles to a canonical turbulent boundary layer profile. For the riblet case, modified Clauser technique [3] is used, assuming a universal gradient in the logarithmic region with a modified or adjusted intercept ΔU^+ (see equation 2). Note that the *A* here refers to the intercept of the smooth wall log law and should not be confused with the amplitude of the meandering riblets defined previously.

$$U^{+}(z) = \frac{1}{\kappa} ln(\hat{z}^{+}) + A - \Delta U^{+}$$
(2)

An upward shift in the mean velocity profile (a negative ΔU^+ or negative roughness function) indicates a drag reduction. Here $U^+ = U/U_{\tau}$ and $\hat{z}^+ = \hat{z}U_{\tau}/\nu$ where \hat{z} is the wall-normal distance from the virtual origin ($\hat{z} = z - z_0$, where z is the measured wall-normal distance from the trough of the riblet geometry and z_0 is an unknown roughness offset). The universal logarithmic constants used here are $\kappa = 0.41$ and A = 5.0.

The measured smooth wall profiles at all three Reynolds numbers are first fitted to the logarithmic region equation to obtain an estimate for the friction velocity U_{τ} (the Clauser technique [3]). Choi [2] suggests that the mean velocity profile over straight riblets obeys the universal logarithmic form, where it was reported that the Clauser plot yields a $-\Delta U^+$ (upward shift) indicating drag reduction. This assumption was applied here. The data are fitted to the modified Clauser equation given in equation (2). This equation alone is difficult to fit to, since there are three unknowns (U_{τ} , z_0 and ΔU^+). There are multiple combinations of these three variables that give a good fit of the data to equation (2), and a unique solution is not obvious. The velocity defect plot for the outer region ($z^+ \ge 100$) is also analysed,

$$\frac{U - U_{\infty}}{U_{\tau}} = f\left(\frac{\hat{z}}{\delta}\right) \tag{3}$$

Outer layer similarity would suggest that when scaled in this manner, the smooth and rough (riblet) data should collapse. Equation (3) offers a further check of the possible combinations of U_{τ} and *e* suggested from equation (2), and regression fitting will yield the most likely candidate combination.

Ultimately, the above methodology of fitting to determine U_{τ} for the riblet surfaces has not yielded conclusive and repeatable results, with any measured change of U_{τ} within the margin of experimental error. For the meandering riblets, we observe that the assumption of outer layer similarity (and hence the use of



Figure 2: Freestream normalised RMS turbulence intensities u'/U_{∞} for smooth wall, meandering and straight riblet surfaces at (a) $Re_{\tau} = 1400$, (b) $Re_{\tau} = 2000$ and (c) $Re_{\tau} = 2800$.

the velocity defect plot) is not entirely justified, with some differences observed in the wake profile for the meandering case. Without this assumption, it is impossible to accurately determine U_{τ} with the current experimental set-up. A drag balance will ultimately need to be implemented in future studies to obtain a direct measurement of U_{τ} . To the best we can determine with the above methodology, we note that meandering riblets appear to behave as a marginally transionally rough surface, i.e. a possible 1-2% drag increase compared to the smooth wall. Straight riblets tend to yield a slight drag reduction (as would be expected from the wealth of literature on these surfaces).

In the absence of accurate and reliable estimates of U_{τ} , the turbulence statistics normalised by the freestream velocity U_{∞} and the boundary layer thickness δ is presented here for comparison between flow over smooth surface and the straight and meandering riblet cases. Figure 2 shows the root-mean-squared turbulence intensity of streamwise velocity fluctuations (u') for the 3 different Reynolds numbers ($Re_{\tau} = 1400$, 2000 and 2800 corresponding to freestream velocity $U_{\infty} = 10, 15$ and 20 ms⁻¹). It is clear from figure 2 that the riblets attenuate the near-wall peak of the turbulence intensity profile, and the effect is more significant for the meandering riblets as compared to the straight riblets as the Reynolds number (and hence h^+ of the riblets) increases. At $Re_{\tau} = 2800$ the near-wall peak intensity for the meandering riblets is attenuated to such an extent that the peak is absent altogether. In making this observation however, it is important to remember that the meandering riblets are of slightly larger riblet height h^+ and spacing s^+ than the straight riblets.

Further from the surface, the intensity over meandering riblets starts to exceed that of smooth wall for $z/\delta \gtrsim 0.02$ or $z^+ \gtrsim 55$, with a peak excess energy occurring at $z/\delta \approx 0.2$ and finally converging with the smooth wall profile at the edge of boundary layer. This may suggest that some of the energy from near-wall region (below $z/\delta = 0.02$) has been shifted to outer region by the meandering riblet. This effect is also observable (to a lesser extent) at $Re_{\tau} = 2000$. This change in shape of the turbulence intensity profiles further verifies our earlier observation that the meandering riblets alter the boundary layer profiles such that the assumption of outer-layer similarity is no longer satisfied.

Premultiplied Energy Spectrum

Figure 3 presents the pre-multiplied energy spectra $k_x \phi_{uu}$ (where k_x is the streamwise wavenumber and ϕ_{uu} is the energy spectrum of the streamwise velocity fluctuations) as function of streamwise wavelength λ_x (= $2\pi/k_x$) and distance from the wall z. The spectra maps are scaled with freestream velocity U_{∞} and the boundary layer thickness δ for comparison to the smooth wall. In figure 3 the pre-multiplied energy spectrum throughout the boundary layer for Reynolds Number $Re_{\tau} = 2000$ (top plots a, b, c) and 2800 (bottom plots d, e, f) are presented. The results for $Re_{\tau} = 1400$ are not presented here since there is very little observable difference between the smooth and the riblet cases (the small viscous-scaled riblet height at this Reynolds Number is insufficient to significantly perturb the energy profile). Figure 3 (a) and (d) show the smooth wall spectra, while (b) and (e) and plots (c) and (f) show the meandering and straight riblet spectra respectively.

In Figure 3 (b) and (e), the horizontal lines plotted on top of the spectra contours represent the scale of the normalized streamwise sinusoidal wavelength Λ_x/δ of the meandering riblet pattern. We can clearly see that close to the wall $(z/\delta \lesssim 0.02)$ at both Reynolds numbers the meandering riblets have significantly reduced the magnitude of the energy contributed by structures of scale $\lambda_x \geq \Lambda_x$ when compared to the smooth wall. Equally significant, for the highest Reynolds number $Re_{\tau} = 2800$, the large-scale energy at the outer peak, centered around $z/\delta = 0.07$ and $\lambda_x/\delta = 6$, is greater in magnitude over the meandering riblets as compared to both the smooth wall and the straight riblets. This outer peak is typically associated with the very large scale motions or 'superstructures' [5], and the implication here seems to be that the meandering riblet geometry is somehow interacting with these very large-scale coherent motions in a manner that increases the overall turbulent energy at this scale. The straight riblets exhibit no discernable change in energy at this outer peak location. This finding is consistent with the turbulent intensity results of figure 2, and confirms that the increased broadband intensity for $z/\delta \gtrsim 0.02$ is due to increased energy in the very long wavelengths. In general we observe that at higher Reynolds Number ($Re_{\tau} = 2800$) the noted effects of the meandering riblets are more pronounced on the energy spectra. At this Reynolds number, the height and spacing of the meandering riblets are larger in wall units, suggesting that a plausible passive periodic forcing is imposed on the boundary layer due to the meandering waves of the riblet pattern. In the near-wall region, the fact that the meandering riblets attenuate energy contributions from scales greater than Λ_x could be interpreted as the result of a spatial periodic forcing and assumed to be a direct consequence of the meandering wavelength. However, an equally plausible suggestion would



Figure 3: Premultiplied energy spectra $k_x \phi_{uu}/U_{\infty}^2$ contours for (a,d) smooth wall (b,e) meandering and (c,f) straight riblets at two Reynolds Number as indicated, plotted as a function of wall normal position z/δ and energetic streamwise length scale λ_x/δ .

be that the meandering riblets have substantially reduced the energy from the near-wall cycle, which has been shown to have a dominant wavelength ($\lambda_x^+ = 1000$)[5] which is very close to the meandering wavelength (Λ_x) of the surface used here. Further tests with vastly different meandering wavelengths could potentially resolve this question.

Conclusions

The meandering riblets significantly perturb the turbulence intensity and premultiplied energy spectrum profiles at high Reynolds Number $Re_{\tau} = 2800$, where the riblet grooves are the largest in viscous wall units. The near-wall peak of the turbulence intensity profile is found to be heavily attenuated, while an increase in the intensity is found in the outer region. This result is further investigated through the premultiplied energy spectra. The near-wall energy contribution from structures of scales greater than the meandering riblet wavelength have been significantly reduced. This could be a result of forcing at the scale of the meandering wavelength, or could equally well be indicative of a more general disruption of the near-wall cycle. More intriguingly, the outer energetic peak is significantly strengthened for the meandering riblets, particularly for the highest Reynolds number. This peak is typically associated with the very largest scale motions (or superstructures), and implies that the meandering pattern, despite the very small roughness height, perturbs the boundary layer profile in such a way that it strengthens coherent motions that exist in the log and wake regions.

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